

EUROPEAN PATENT SPECIFICATION

- (13)** Date of publication of patent specification: 16.08.89
(21) Application number: 83300324.7
(22) Date of filing: 21.01.83
(8) Divisional application 87114727 filed on 08.10.87.
(51) Int. Cl.⁴: G 05 D 1/00, B 64 C 13/46,
B 60 K 26/02

(54) Control apparatus.

(39) Priority: 22.01.82 GB 8201907
22.10.82 GB 8230312

(43) Date of publication of application:
10.08.83 Bulletin 83/32

(45) Publication of the grant of the patent:
16.08.89 Bulletin 89/33

(84) Designated Contracting States:
DE FR IT NL SE

(5) References cited:
DE-A-2 723 562
FR-A-1 119 236
GB-A-1 549 178
GB-A-2 073 887
US-A-2 630 284

(73) Proprietor: BRITISH AEROSPACE PUBLIC
LIMITED COMPANY
11 Strand
London WC2N 5JT (GB)

(72) Inventor: Hill, Graham Arthur
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)
Inventor: Acklam, Donald John
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)
Inventor: Carr, Brian
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)
Inventor: Latham, Phillip Anthony
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)
Inventor: Bird, David
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)
Inventor: Kaye, Arthur
British Aerospace P.L.C. Warton
Preston, Lancs. PR4 1AX (GB)

(74) Representative: Dowler, Edward Charles et al
British Aerospace Public Limited Company
Corporate Patents Department Brooklands Road
Weybridge Surrey, KT13 OSJ (GB)

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Courier Press, Leamington Spa, England.

EP 0 085 518 B1

SON 0010695

Description

This invention relates to certain developments and improvements in control devices of the kind disclosed in our UK Patent Application No. 2,073,887A, which control devices enables an operator to control associated apparatus to which, in use, the control device is connected. By way of example, the control device may be a thrust demand unit ('throttle box') for controlling an engine of an aircraft. Amongst others, UK Patent Application No. 2,073,887A discloses embodiments of control devices wherein the movement of a manually movable control member effects a control action on the associated apparatus, via suitable output means such as a position encoder and electrical signal link or a mechanical coupling, and wherein an electrical stepper motor is connected to the control member and is so energised by a computer controlled motor drive that a noticeable increase in the resistance to manual movement of the control member is generated by the motor when the control member is at one or more pre-set detent positions in its overall range of movement. In addition or as an alternative, the computer may control the motor drive system so as to generate a controlled feel force resisting the manual movement of the control member throughout its range of movement. As well as the above, the control device may be operable for being set into an automatic control mode in which the computer drives the stepper motor to produce motor driven movement of the control member.

It is often desirable to vary the position at which detents occur depending on some parameter associated with the apparatus being controlled. Examples of variable mechanical detents are given in German Patent 2723562 and in UK Patent 1,549,178 to produce increase throttle pedal resistance when engine speed or fuel consumption exceed pre-set limits.

It is an object of the present invention to provide a control device and a method of operating it which includes a variable, electrically generated detent position.

According to one aspect of the present invention, there is provided a method for the operation of a control device comprising a manually movable control member (5, 35, 55, 200, 301), output means (2, 14, 33, 52, 302, 402) for exerting a control action on associated equipment in dependence upon the movement of the control member, and a computer driven electrical stepper motor (6, 31, 51, 306) coupled to the control member, the method including providing the computer (1, 101, 318, 406) with information about the position of said control member within its range of movement and using the computer to control the energisation of said motor so as to generate a noticeable increase in resistance to the manual movement of the control member through at least one detent position in said range, the method being characterised in that the computer is further provided with information about one or more physical variables affecting said associated equipment and the computer uses

said information to calculate the location of said detent position such that said location changes during use of the control device in dependence upon changes in said one or more variables.

According to a second aspect, there is provided a control device for controlling the operation of associated equipment, for example apparatus on board an aircraft or road vehicle, the control device comprising a manually movable control member (5, 35, 55, 200, 301), an electrical stepper motor (6, 31, 51, 306) coupled to the control member, a sensor (18, 102, 12, 33, 52, 302) for sensing the position of the control member within its overall range of movement, a computer (1, 101, 318, 406) connected to the sensor and the stepper motor for energising the motor to produce a noticeable increase in resistance to said manual movement of the control member through at least one discrete detent position of the control member, and output means (2, 14, 33, 52, 302, 402) for transmitting a control action to said associated equipment dependent upon the position of the control member, characterised in that the control device further comprises signal supply means (103) for supplying to the computer (1, 101, 318, 406) signals indicative of at least one physical variable affecting said associated equipment and said computer is programmed to respond to said signals to vary the location of said detent position in dependence upon said physical variable.

According to a third aspect, there is provided a motor driven vehicle comprising a control device in accordance with the second aspect as described above, the control device being connected to control the operation of speed varying apparatus on board the vehicle, said signal supply means comprising transducer means for forming signals representing at least one physical variable which is linked with the operating economy of the vehicle, and the computer is programmed to calculate from said signals a setting of said speed varying apparatus optimised for said economy and to correspondingly locate said detent position in the range of movement of the control member.

According to a fourth aspect of the invention, there is provided an aircraft comprising a control device in accordance with the second aspect described above, the control device being connected to control the throttle setting of an engine of the aircraft, said signal supply means being operable to supply a signal indicative of aircraft altitude, and the computer being programmed to vary the location of said detent in dependence upon said altitude.

For a better understanding of the invention, reference will now be made by way of example to the accompanying drawings, in which:—

Figure 1 is a diagrammatic view of an accelerator control for a motor vehicle, the control being coupled to a device for generating an advisory detent position in the movement of the control,

Figure 2 is a sectional view of the detent force generating device of Figure 1,

Figure 3 is a block diagram of a system for generating one or more detents and/or a feel force

in association with the movement of a control member, and

Figure 4 is a simplified diagram of part of an aircraft throttle box.

Figures 5 and 6 are a sectional plan and a sectional elevation of a control device,

Figures 7, 8, 9 and 10 are diagrammatic views of respective further control devices, and

Figures 11, 12 and 13 are simplified circuit diagrams of respective drive/interface circuits for use in conjunction with control devices.

Referring to Figure 1, a motor vehicle may comprise a series of transducers (not shown) for forming electrical signals indicative of respective operating conditions of the vehicle and a signal processing device 1 for receiving those signals and using them to determine a vehicle control setting which, given the relevant conditions, optimises some predetermined further operating parameter. Thus, in the illustrated case, the transducers indicate the road speed of the vehicle and the speed of its engine, the selected driving gear, engine and air temperature, and the pressure of the air or fuel/air mixture within the engine inlet manifold. The signal processing device, which may comprise a microprocessor based computer system, determines from these parameters a throttle setting to give maximum fuel economy. By way of example, the signal processing device could be operable, in dependence upon the signal it receives, to select an accelerator setting value from a stored set of such values, i.e. a kind of look-up table, the stored values having been derived by separate experimentation with the same or a similar vehicle and engine over appropriate ranges and combinations of the likely operating conditions. The signal processing device may be operable simply to select the most nearly appropriate value from the look-up table or it may be operable to interpolate between these values in ways known in connection with look-up table operation of computer systems. Alternatively, in cases where fuel economy is linked to values of the transduced operating conditions and the accelerator setting by electronically calculable functions, the signal processing device may be operable to calculate the optimum accelerator setting directly from the transduced conditions, or possibly the device may operate by a combination of direct calculation and reference to a look-up table.

The accelerator control shown comprises the usual foot pedal 2 connected by say a Bowden cable (not shown) to the engine speed control device, for example to a carburettor valve. The pedal is supported by a pivot mounting 3 and, pivotably connected to the pedal near the foot engaging portion 4 thereof, is one end of shaft 5 by which the pedal is coupled to a detent force generating device 6.

Referring to Figures 1 and 2 together, the device 6 comprises an elongate box-like housing 7 which has two lugs 8 extending from one side thereof at opposite ends of the housing and of which one end is pivotably connected to the cabin/engine

compartment bulkhead 21 of the vehicle. The shaft 5 is mounted in two bearing bushes 9 supported by respective ones of the lugs and is able to move in the direction of its axis in response to movement of the accelerator pedal.

Within the housing 7 are contained the stator 10 and mover 11 of a linear stepper motor of the kind disclosed in our UK patent application No. 2,073,887A. The stator 10 consists of an elongate cylindrical member 12 made of magnetisable material and formed with a series of annular recesses to leave defined intervening pole teeth 13. The recesses are filled with non-magnetic material 14, such as copper, and the whole is covered by a thin plating 15 of material having good sliding properties, chromium for example. Meanwhile, the mover 11 comprises a cylindrical housing within which is mounted an electro-magnetic pole assembly. This pole assembly is not shown but, as described in patent application No. 2,073,887A to which attention is directed, it may comprise two ring-shaped poles each carrying a winding and each having two spaced sets of annular grooves in its internal surface to define respective sets of pole teeth. Between the poles is a magnet assembly including a rare-earth permanent magnet. The pitch and spacing of the mover pole tooth sets and the stator pole tooth pitch are such that, by properly sequencing the energisation of the mover windings, the mover can be stepped along the stator between adjacent ones of a series of positions at which the reluctance to flow of magnetic flux energy between mover and stator is a minimum. While the mover is at one of these minimum reluctance positions, the force needed to move it out of that position depends on the level of energisation applied to the windings. Thus the resistance to manual movement of the mover can be varied to give a desired level of 'feel' in that movement and/or, when the mover is being manually moved through one or more predetermined positions in its range, the winding energisation can be increased from a normal standing level so as to generate a marked resistance or 'detent' at that or those positions.

Connected to the mover is an actuator arm 16 which extends from an elongate slot in the housing wall between the lugs 8 and is there attached to shaft 5. A compression spring 17 is engaged around shaft 5 to act between arm 16 and that one of the lugs 8 which is furthest from the pedal 2.

Within the housing 7, the mover 11 is coupled to a position encoder 18 which, in this case, takes the form of a potentiometer with a resistive and a conductive track formed on a fixed substrate 19 and a bridging contact 20 movable with the mover 11 so as to bridge the tracks at a position dependent upon the position of the mover and thereby give a resistance value between the track end terminals which is also dependent upon the mover position. The encoder may have more than one pair of tracks and bridging contact in order to give a degree of redundancy. Alternatively, any other suitable form of position encoder may be

used, for example any of those disclosed in UK patent application No. 2,073,887A.

The device 1 is constructed and/or programmed to compare the instant position of the mover 11, and hence in effect the instant position of pedal 2, with the optimum position determined from the transduced operating conditions as described earlier. On the basis of the comparison, the device 1 so controls the energisation of the windings of mover 11 that depression of the pedal down to the optimum position is comparatively easy while movement past this point is met by a resistance which, although substantially increased, can nevertheless be overcome if desired.

It will of course be appreciated that the point at which the increased resistance is felt is not fixed as would be the case if some simple mechanical detent system were coupled to the accelerator pedal. Rather the detent position with the described apparatus is varied so as to indicate the optimum accelerator position or optimum maximum accelerator for a range of the other operating conditions.

If desired, the described apparatus can be provided with additional facilities such as the ability to maintain a particular accelerator position. For this, a driver control switch 22 is provided and, when operated, this instructs the device 1 to increase the winding energisation of mover 11 to a level at which the pedal 2 is held in position against the return force of spring 17 and any other springs which may be included in the accelerator control linkage.

Instead of just holding the pedal position, the processor device 1 could act to maintain the road speed of the vehicle constant. For this, the device 1 could register the transduced signal representing road speed at the time when control switch 22 is operated and then, on the basis of a comparison between the registered and present value of the road speed signal, could so energise the windings of mover 11 when necessary that the mover is actually driven along the stator, the position of pedal 2 being thereby adjusted to maintain road speed constant. While a constant road speed mode is selected, the device 1 could be operable, on the basis of engine speed for example, to sense the desirability of selecting a lower gear sometimes in order to hold the desired speed, and to indicate this to the driver via a suitable warning indicator 23, for example an audible warning device and/or a lamp. If desired, override switch means 24 can be arranged to sense any manual operation of one or more of the clutch, brake and accelerator controls of the vehicle and to then initiate overriding disengagement of the speed holding mode of device 1 whatever the state of switch 22.

The illustrated apparatus could be used for road vehicles as described but also for railway locomotives, particularly advantageously for road or rail vehicles having automatic transmissions (in which case the gear selection warning device may not be needed). The apparatus could also be used for boats and even aircraft. In the case of those

propellor, fan or screw driven vehicles where speed control is by varying the blade pitch while the engine speed is automatically kept constant, the detent force generating device 6 would of course be coupled to the pitch control.

The transduced parameters fed to device 1 are of course given only by way of example as is the parameter to be optimised. The transduced parameters depend upon the parameter to be optimised and, even for a particular parameter, depend upon the kind of vehicle and may also be a matter of choice, made on an economic basis for example. Thus, some indication of a maximum economy accelerator position can be obtained from the inlet manifold pressure alone of a car engine and hence a simplified version of the illustrated apparatus may comprise only a transducer for this parameter.

Instead of the linear stepper motor shown, the detent force generating device 6 could comprise a rotary stepping motor connected via a rotary/linear motion converter to shaft 5 or coupled to the pedal 2 without rotary/linear conversion, for example directly to the pedal at its pivot point.

The invention is not limited to speed control of a vehicle. For example, in figure 1, the pedal 2 could be the clutch control pedal of a vehicle instead of the accelerator pedal and the device 1 could sense the actual degree of engagement of the clutch itself and so energise the detent generator 6 as to encourage operation of the pedal in a way which reduces wear and tear of the clutch. This would be particularly useful in a large vehicle such as a rear-engined bus with a remote control clutch.

The electric stepper motor which generates the desired movement resistance and detents and which may also be operable to drive the movement of a movable member, may be incorporated in an overall system such as that shown in figure 3 where reference 100 indicates the feel unit and an energisation controller 101 is operable to supply driving/control signals for this unit on the basis of information received by the controller both from a position sensing device 102 coupled to the feel unit and from a suitable input/output interface unit 103 which interfaces the overall system to external influences. Thus, the position at which the moving member of the feel unit 100 in relation to the stationary member feels a detent force, and the magnitude of that force, may be controlled by the motor controller 101. The detent force may operate for a defined proportion of the available travel of the moving member at the same or a varying force level. Signals from external sources are sent to the electronic interface unit 103 and operated upon according to requirement, to produce signals for the motor control unit 101 which will define the magnitude, position and characteristic form of the detent. For example in the case of an aircraft throttle box, the position of say a "MAX DRY" detent i.e. a throttle position at which there is attained maximum thrust without say any afterburning system or other thrust supplement system in operation, can be varied as a function of altitude.

The controller 101 can be made operable to cause the base level "friction" force that resists the motion of the moving member to be increased when the throttle handle is in an idle position and signals are received at the interface circuit indicating that the aircraft undercarriage is down and aircraft weight is actually on the ground. By such a means it can be ensured that if, on landing, the aircraft is subject to sudden deceleration, the throttle handles are not thrown forward. It will be realised that the detent position(s) and feel force can be changed continuously as a function of vehicle speed, altitude or any other desired parameters or combination of parameters.

In the apparatus of figures 1 and 2 the mover 11 of the stepper motor has the electromagnetic windings and so power to drive the motor is applied to the mover by way of flexible wiring. To avoid the need for such wiring and possible damage thereto consequent upon its bending as the mover moves, the motor could be adapted by incorporating the electromagnetic windings in the stator rather than the mover and therefore allow the drive power to be applied to the stator.

The handle or grip portion of an aircraft throttle-box may incorporate control switches, for example a switch for operating a reverse-thrust mechanism of the aircraft. In this case, and any other where switches and/or indicators are incorporated in or connected to the movable control member of the control device, the need for wiring coupled to the movable member may be removed by use of a wireless data transmission system between the mover and fixed data processing circuitry incorporated in the unit, for example mounted on the "stator" of the unit. Thus, as shown in Figure 4, the switch information could be encoded by circuit 104 onto an AC signal which is fed into a coil 105 mounted in the moving member and wound around magnetic bar 106 along which the moving member travels. Another coil 107 attached to the stationary member would also be wound around the magnetic bar in a fixed position and would receive the encoded alternating signal and pass it to the data processing circuitry 108 for analysis.

Data could be passed either from the moving member to the stationary member or from the stationary to the moving member in a multiplexed manner. Power for the circuitry in the moving member could be derived from the data transmission signal used to communicate from stationary to moving member and, with the use of very low power electronic circuitry, could provide sufficient power during the communication between moving and stationary member. The power in the moving member would be replenished during the next transmission from stationary to moving member.

It may be desirable for the control device such as a microprocessor based computer system which is controlling the stepper motor to receive an indication that the mover of the motor has in fact executed a step or series of steps com-

manded by the control device. For this purpose, application No. 2,073,887A disclosed the idea of comparing the e.m.f. appearing across each motor winding with a waveform corresponding to what would appear across the motor winding if the commanded step were executed (this waveform is different to that which appears if the step is not executed because of the influence of movement-induced e.m.f.).

As a further possibility, the execution or otherwise of a commanded step can be detected by coupling a strain-gauge to the motor stator or incorporating such a gauge in the mover so as to sense the small strains and extraneous movements which occur in the motor elements during its operation.

Figures 5 and 6 of the accompanying drawings show in diagrammatic form a control device of the kind described and claimed in our patent application EP-A-0265738 and which could form the basis of say an aircraft throttle-box. The device comprises a box 1 which, in plan, is rectangular and of which the top is curved with respect to axis 3 and has a longitudinally extending slot 4 in it. Mounted to shaft 5a for pivoting movement about axis 3 is a lever 5 which extends out through slot 4 and is there fitted with a grip handle or knob 6. Thus, the lever can be moved through the range of which the extremes are shown by the chain-lined representation of the lever.

Mounted within box 1 next to lever 5 is the stator rod 7 of a linear stepper motor of the kind disclosed and illustrated in our patent application No. 2,073,887A. The rod is fixed by having its ends 8, which are stepped, entered into respective holes in the ends of the box. The moving element or 'mover' 9 of the stepper motor has a projecting peg 10 which extends into a slot 11 in the lever 5. Thus, as the lever 5 is manually moved, the mover 9 moves with it along rod 7 and, if the mover 9 is driven along rod 7, it moves the lever 5. As in application No. 2,073,887A, during manual movement of lever 5, the stepper motor may be operable to provide a fixed, variable or programmed degree of 'feel' to the manual movement.

In order to transmit the control action to whatever apparatus is being controlled, e.g. the aircraft engine controller if the device shown is a throttle-box, an electronic or mechanical linkage or both can be provided. The device shown has both. The electronic linkage comprises any suitable kind of linear position encoder fixed in box 1 and including a fixed element 12 and a movable element 13 coupled to motor mover 9. By way of example, the encoder could be an Inductosyn (Registered Trade Mark) type of encoder, a linear potentiometer, or a light-operated type of encoder. The mechanical back-up linkage includes a push-pull rod 14 pivotably coupled to lever 5, the pivot pin 15 extending through a slotted hole 16 in lever 5.

As an alternative to what is shown, the position encoder could be a rotary position encoder

coupled to the lever 5, directly or *via* say a gear arrangement, at the pivot point 3. The mechanical linkage could also comprise any form of rotary coupling, say a gear drive belt and pulley, or a reel which winds up a Bowden cable coupled to lever 5 *via* its supporting shaft 5a.

The device shown, with suitable modification possibly, could be used for various control applications. For example, it could be adapted for foot-operation and then be operable for controlling the accelerator of a road vehicle.

Figures 7, 8 and 9 of the drawings show respective further control devices in diagrammatic form. In Figure 7, a box 30 with a slotted top contains a rotary stepper motor 31 of which the rotor (not shown) is coupled to one end of a ball screw 32. The other end of the screw 32 is coupled to the input shaft of a rotary position encoder 33 and the screw carries, engaged with the screw so that it moves back and forth along the screw when the latter rotates, a rider member 34 having a grip lever 35 extending therefrom and out of the slot in the top of the box. As before then, motor 31 can rotate to move lever 35 and to exert the desired control action *via* encoder 33, the operation of the motor nevertheless being manually overridable by forced manual movement of the lever. Also, when the lever is being manually moved normally, the motor can provide the required degree or variation of feel to that movement.

Figure 8 is similar to figure 7 except for the slight modification whereby the motor 31 is positioned below screw 32 and is coupled thereto *via* a gear train 40 and the encoder 33 is coupled directly to the motor shaft.

A ball screw, which is a known item can have a very low coefficient of friction and is 'reversible' (i.e. if rotated it moves the rider and, if the rider is moved, the screw is rotated) even for quite fine pitches of the screw. Hence, such a screw is well suited to use in a control device as shown. The encoder used can be a multiturn encoder suitably chosen from say the range supplied by the firm Muirhead Vactric.

As will be realised, the devices shown in figures 7 and 8 can be configured in various ways whilst still retaining the feature of a rotary motor and screw rod, preferably a ball-screw, coupled to a rider which is in turn coupled to a manual operating lever or knob. Figure 9 shows a box 50, again with a slotted top, and containing a rotary stepper motor 51 and a rotary position encoder 52 each coupled to a toothed-belt wheel 53. A toothed belt 54 is entrained round the wheels 52 and couples them together and the belt is also coupled to a rider 55 from which a manual operating lever 56 extends through the slot in box 50.

Using rotary motors and encoders is considered better than using linear devices because thereby it may be easier to carry out proper dirt and moisture exclusion sealing in the various components and the device as a whole. Also by using 'multiturn' rotary components, resolution

and such may be able to be improved and it may be easier, for example by proper use of gears, ball-screws and such, to obtain mechanical advantage enabling motors and such to be smaller or less powerful.

The control device of figure 10 comprises a lever 200 of which one end protrudes from a housing 201 and is formed with a grip portion 203 and of which the other end, inside the housing, has affixed to it the rotor or mover part 204 of a segmental stepper motor, which motor also comprises an arcuate stator 205. Respective curved surfaces 206 of the mover 204 and stator 205 face each other and are formed with pole teeth and the lever 200 is pivoted for movement about an axis 207 at the centre of curvature of the surfaces 206. When energised by supply means (not shown), the surfaces 206 move relative to one another or produce a feel force between each other which produces a corresponding movement or feel force for the lever 200.

Figure 11 shows diagrammatically an aircraft throttle box wherein a lever 301 is connected at one end to the shaft of a rotary position encoder 302 so that the lever can pivot about the axis of the encoder shaft. The encoder 302 is preferably of the multiple-channel type so as to give a plurality of position indicative output signals which normally are substantially the same — the object of course being to give a required degree of redundancy. The other end of lever 301 protrudes from the throttle box housing (not shown) and is there fitted with a manual grip portion 303 which, in turn is fitted with a plurality of control switches, for example a switch for bringing in the reverse thrust mechanism of the aircraft as so on. An intermediate portion of the lever 301 is pivotably connected to the 'nut', i.e. the linearly moving female part, 304 of a recirculating ball-screw assembly which also comprises a threaded rod 305 along which the nut moves when rod 305 is rotated. Rod 305 is coupled to a rotary stepper motor 306 *via* gearbox 307. As with previous embodiments described herein, the lever 301 can be moved manually to exert a control action on the engine of an aircraft in which the throttle box is fitted, in which case the stepper motor 306, *via* gearbox 307 and the ball-screw assembly, can be so energised and controlled that it provides a controlled feed force resisting the manual movement and/or controllable or programmable detents i.e. noticeably greater resistance to movement at certain positions within the overall range of movement of the lever. Also, the stepper motor can actually move the lever 301 under the control of the aircraft autopilot.

In order to interface the throttle box to the autopilot computers, the engine speed control computer, and local controllers for the various items to be controlled by the switches on grip portion 303, an interface/control unit 310 is provided. This is powered by its own dual redundant power supply unit 311 from respective independent aircraft power source channels A and B and it comprises a series of digital interface units 312

to 317 and central processor unit 318 which may comprise a microprocessor based computer system. The interface unit 312 interfaces unit 318 to the stepper motor and hence of course is constructed to provide appropriate energisation signals for driving the motor as known in the art, or for providing reduced level energisation signals to produce the aforementioned feel and detent forces. The other interface units handle various control and information signals. Thus, unit 313 is connected between the unit 318 and the grip switches, unit 314 passes the position encoder signals to the unit 318 from encoder 302, unit 315 transfers to the unit 318 from the autopilot or to the autopilot from unit 318 signals indicative of whether the throttle box is in its manual or automatic state of operation, and the interface units 316 and 317 are respectively parallel input and output interfaces which connect the unit 318 to aircraft data buses (not shown), these data buses also communicating with the aircraft autopilot, throttle control computers and local controllers as known in the art.

The central processing unit 318 is programmed to handle the transfer of signals between the various interface units as necessary and to carry out the sequencing of the winding drive signals for the stepper motor to give the required direction and amount of movement. In autopilot mode, when a change in throttle position is required, the aircraft autopilot passes to the unit 318 a demanded absolute position of the throttle control grip. The unit 318 then calculates the amount and direction of movement required to get the grip to that new demanded position from its present position and then outputs the appropriate drive pulse sequence to the motor. It also controls the frequency of the pulse sequence to give a smooth movement, i.e. with an initial increase of frequency up to some predetermined value then a constant frequency period and then a smooth reduction. As a result the movement comprises an initial acceleration phase, an intermediate constant speed phase and then a smooth deceleration to a standstill at the new position. The calculation of the acceleration and deceleration rates and times is done so as to optimise the time taken to reach the new position without exceeding any predetermined maximum rates. The unit 318 also receives from the aircraft computers signals indicating the positions at which there are to be generated detents to manual movement of the control grip, and the level of resistance to movement which is to be encountered at those positions. If the throttle box is in manual mode, the unit 318 maintains a comparison of the actual grip position and the demanded detent position(s) and, if and when the grip is moved through a detent, it initiates an increased motor energisation level to give the desired detent resistance. The demanded detent position is calculated by the aircraft computer on the basis of say instant navigation and operating parameters of the aircraft.

For a throttle box comprising a linear stepper

motor of the kind illustrated in say figure 2 herein, the interface/control unit could be as shown in figure 12. This unit is again energised by a dual redundant power supply 401 and interfaces the throttle box input and output data buses (preferable multiple channel buses to give a degree of redundancy as known in the art) which communicate with the aircraft autopilot and throttle control computers. The throttle box position encoder 402 is shown as a block and can be of any suitable form, e.g. a linear potentiometric type of device as shown in figure 2, and the position indicative signal therefrom is fed to an encoder logic unit 403 which in turn feeds an output data bus interface unit 404 and a direction sensing and acceleration/deceleration control unit 405. The unit 405 is interposed, along with a detent logic unit 406 and force control unit 407 between a parallel input interface 408 for the input data bus and a motor drive logic unit 409 which actually provides the energisation signals for the throttle box stepper motor. The unit 405 provides, in any automatic movement of the motor, acceleration and deceleration phases to ensure smoothness of that movement and also determines the direction in which the motor must move in order to get the grip portion of the throttle box from its instant position to a new position demanded by the aircraft computer via the input bus and interface 408. When in the manual mode, the unit 405 also senses the direction of any manual movement of the grip, i.e. by sensing the direction of change of the position encoder signal, and controls the drive logic 409 to ensure an appropriate energisation sequence of the motor windings either for generating the required resisting feel force or, possibly, if the pilot prefers it, an assisting force which reduces the manual force required to move the grip. The force control unit determines the general level of energisation of the motor windings in response to aircraft computer supplied signals indicative of the general feel force, detent force and/or movement forces required, while the detent logic unit senses when the grip is being manually moved through a detent position (which detent position has been communicated to by the aircraft computer) and then initiates the production of the winding energisation level appropriate to generate the detent.

As will be appreciated, in either of figures 11 and 12, any desired level of redundancy may be incorporated. For example, in figure 11 each output channel of the encoder could be coupled to a respective separate interface/control unit like the one shown, each unit incorporating its own central processor unit.

As shown diagrammatically in figure 13, a potentiometric position encoder for use in an aircraft throttle box comprising a linear stepper motor 524 could have five potentiometers 506 to 510 each consisting of a resistive track and a conductive track bridged by a wiper. The wipers are ganged together to move with the moving element of the linear stepper motor. A reference voltage is applied between the ends of the resis-

tive strip of each potentiometer so that there appears on the corresponding conductive strip an output voltage dependent on the position of the slider and hence on the position of the moving element of the motor.

The reference voltage of each potentiometer is obtained from a duplex supply arrangement comprising two independent reference generators 520 and 521 fed by independent power feeds 522 and 523 of the aircraft. The potentiometers 506 and 507 are fed from the generator 520 while the potentiometers 508 to 510 are fed from generator 521. The output signals DDO from the potentiometers 506 to 509 are led off to an interface/control unit (not shown) which communicates with local engine controllers (not shown). Meanwhile, the output from potentiometer 510 is used as a feedback signal for controlling the motor 524. Accordingly, this signal is digitised by analogue to digital converter 525 and then fed to a control circuit 526 along with a signal APD from the aircraft autopilot (not shown) and representing a level of engine thrust demanded by the autopilot. Using these signals, the control circuit 526 is able to feed trains of drive pulses to the motor 524 via a power output stage 527 of circuit 526 so as to move the throttle lever according to the autopilot demand, each train of pulses preferably including an initial portion in which the pulse repetition rate is ramped up and an end portion where it is ramped down again so that the motor movement comprises a smooth combination of an acceleration phase, a constant-speed phase and a deceleration phase. At the same time, the circuit 526 may receive a signal from a pilot's control device 528 to set the standing energisation level of the motor windings and hence the resistance to manual movement of the throttle lever. Also, the circuit may maintain in store a record of the throttle lever position and, at certain positions, increase the winding energisation so as to provide, electrically, detents at these positions. The circuits may also respond to a predetermined value of signal APD, the value zero for example, indicating that the autopilot is switched off and to then set itself into a manual mode in which it does not output drive pulses to motor 524 but still maintains in store a record of the throttle lever position as the lever is manually moved by the pilot and controls the standing and detent energisation of the windings.

The possibility has been mentioned earlier herein of using, in control apparatus as described, a linear stepper motor in which the electromagnetic drive windings are incorporated in the motor stator rather than the mover. Also, mentioned was the possibility of using some form of wireless transmission, like that shown in figure 4 hereof for example, in order to extract control signals from switches mounted on a movable member or lever of the control apparatus. In each case, the object was to avoid the need for cables to extend from the movable member and/or the motor mover to a fixed point of the apparatus. On the

other hand, as will be readily apparent, the use of such flexible cables because of its simplicity as compared to say wireless transmission may well be substantially advantageous. In this connection, it will be further noticed that embodiments of the invention described herein where the movable control member has an arcuate movement, as shown in figures 5 and 6 or figure 10 or 11 say, are advantageous as compared to the 'linear' embodiments of say figures 7, 8 and 9. The reason for this is that the conductors from the grip mounted control switches and/or from the motor mover can be led down along the movable member and then make the transition across to a fixed part of the apparatus at or near to the point where the movable member is pivotably mounted. At this point, the amount of movement which has to be taken up and hence also the length of the free or 'flying' portion of each conductor, is much less. Thus, as shown in figure 6, a winding supply cable 601 from the motor mover 9 can have a short 'free' portion 602 extending between the mover 9 and the member 5, then a portion which is fixed to or extends inside the member 9, and then a further 'free' portion 603 extending from the member 9, reasonably close to the pivot axis 3, over to an exit aperture in the base of housing 2. One or more further cables (not shown) could extend along the member 5 between the exit aperture and control switches (not shown) mounted on or near the grip portion 6.

Claims

1. A method for the operation of a control device comprising a manually movable control member (5, 35, 55, 200, 301), output means (2, 14, 33, 52, 302, 402) for exerting a control action on associated equipment in dependent upon the movement of the control member, and a computer driven electrical stepper motor (6, 31, 51, 306) coupled to the control member, the method including providing the computer (1, 101, 318, 406) with information about the position of said control member within its range of movement and using the computer to control the energisation of said motor so as to generate a noticeable increase in resistance to the manual movement of the control member through at least one detent position in said range, the method being characterised in that the computer is further provided with information about one or more physical variables affecting said associated equipment and the computer uses said information to calculate the location of said detent position such that said location changes during use of the control device in dependence upon changes in said one or more variables.

2. A control device for controlling the operation of associated equipment, for example apparatus on board an aircraft or road vehicle, the control device comprising a manually movable control member (5, 35, 55, 200, 301), an electrical stepper motor (6, 31, 51, 306) coupled to the control

member, a sensor (18, 102, 12, 33, 52, 302) for sensing the position of the control member within its overall range of movement, a computer (1, 101, 318, 406) connected to the sensor and the stepper motor for energising the motor to produce a noticeable increase in resistance to said manual movement of the control member through at least one discrete detent position of the control member, and output means (2, 14, 33, 52, 302, 402) for transmitting a control action to said associated equipment dependent upon the position of the control member, characterised in that the control device further comprises signal supply means (103) for supplying to the computer (1, 101, 318, 406) signals indicative of at least one physical variable affecting said associated equipment and said computer is programmed to respond to said signals to vary the location of said detent position in dependence upon said physical variable.

3. A motor driven vehicle comprising a control device according to claim 2, the control device being connected to control the operation of speed varying apparatus on board the vehicle, said signal supply means comprising transducer means for forming signals representing at least one physical variable which is linked with the operating economy of the vehicle, and the computer is programmed to calculate from said signals a setting of said speed varying apparatus optimised for said economy and to correspondingly locate said detent position in the range of movement of the control member.

4. A vehicle according to claim 3, wherein said control member comprises a foot operated accelerator pedal (4), the speed varying apparatus is a throttle controlling the supply of fuel to an internal combustion engine, and said setting of the speed varying apparatus is calculated for optimum fuel economy (figures 1, 2 and 3).

5. A vehicle according to claim 3 or 4, including switch means (22) connected to the computer (1) and operable for causing the computer to hold said control member at a position in which it has then been set by said operator.

6. A vehicle according to claim 3 or 4, wherein said computer is further operable for energising said motor so as to produce motor-driven movement of said control member, said parameter indicative signal supply means includes means for sensing the speed of said vehicle, and the control device includes switch means connected to the computer means and operable for causing the computer means to vary the position of the control member to maintain said vehicle speed at a constant value.

7. An aircraft comprising a control device according to claim 2, the control device being connected to control the throttle setting of an engine of the aircraft, said signal supply means being operable to supply a signal indicative of aircraft altitude, and the computer being programmed to vary the location of said detent in dependence upon said altitude.

8. An aircraft according to claim 7, said para-

meter indicative signal supply means being further operable to supply a signal indicating that the aircraft has landed and the computer being programmed to respond to the aircraft landed signal by increasing the energisation of the motor to lessen the possibility of said control member then inadvertently moving away from a low throttle-setting position thereof.

9. An aircraft according to claim 7 or 8, wherein said computer comprises an input connected to an autopilot computer on board the aircraft and is alternatively operable in response to signals received from said autopilot computer to energise said motor in a manner which produces movement of its mover part and of said control member.

10. An aircraft according to any one of claims 7 to 9, including pilot controllable switch means supported by said control member, switch signal receiving means (108) fixed with respect to the control device, and wireless transmission means (104, 105, 107) for transmitting signals indicative of the state of said switch means to the receiving means (Figure 4).

Patentansprüche

1. Verfahren zur Betätigung einer Steuervorrichtung, die ein manuell betätigbares Steuerglied (5, 35, 55, 200, 301), ein Abtriebsglied (2, 14, 33, 52, 302, 402) zum Ausüben einer Steuervirkung auf eine zugeordnete Anlage gemäß der Bewegung des Steuergliedes und einen durch einen Computer gesteuerten elektrischen Schrittmotor (6, 31, 51, 306) aufweist, der mit dem Steuerglied gekoppelt ist, wobei das Verfahren die folgenden Schritte umfaßt: der Computer (1, 101, 318, 406) wird mit einer Information über die Lage des Steuergliedes innerhalb seines Bewegungsbereiches informiert und es wird der Computer benutzt, um die Erregung des Motors derart zu steuern, daß eine merkbare Erhöhung des Widerstandes gegenüber der manuellen Bewegung des Steuergliedes über wenigstens eine Hemmstelle in diesem Bereich erzeugt wird, dadurch gekennzeichnet, daß der Computer außerdem mit Informationen über eine oder mehrere physikalische Variable versehen wird, die die zugeordnete Anlage beeinflussen und daß der Computer diese Information benutzt, um die Lage der Hemmstelle zu berechnen, derart, daß diese Stelle sich während der Benutzung der Steuervorrichtung in Abhängigkeit von Änderungen der besagten einen oder mehreren Variablen ändert.

2. Steuervorrichtung zur Steuerung der Arbeitsweise einer zugeordneten Anlage, beispielsweise eines Gerätes an Bord eines Flugzeugs oder eines Straßenfahrzeugs, wobei die Steuervorrichtung folgende Teile umfaßt: ein manuell betätigbares Steuerglied (5, 35, 200, 301), einen elektrischen Schrittmotor (6, 31, 51, 306), der mit dem Steuerglied gekuppelt ist, einen Sensor (18, 102, 12, 33, 52, 302) zur Feststellung der Lage des Steuergliedes über dem Gesamtbewegungsbereich, einen Computer (1, 101, 318, 406), der mit dem Sensor

und dem Schrittmotor verbunden ist, um den Motor zu erregen und ein beträchtliches Ansteigen des Widerstandes gegenüber einer manuellen Bewegung des Steuergliedes über wenigstens eins diskrete Hemmstelle des Steuergliedes zu erzeugen und einen Abtriebsglied (2, 14, 33, 52, 302, 402) zur Übertragung einer Steuerungwirkung auf die zugeordnete Anlage in Abhängigkeit von der Stellung des Steuergliedes, dadurch gekennzeichnet, daß die Steuervorrichtung außerdem eine Signalquelle (103) aufweist, um dem Computer (1, 101, 318, 406) Signale zu liefern, die wenigstens eine physikalische Variable kennzeichnen, die die zugeordnete Anlage beeinflußt, und daß der Computer so programmiert ist, daß er auf diese Signale anspricht, um die Lage der Hemmstelle in Abhängigkeit von der physikalischen Variablen zu ändern.

3. Motorgetriebenes Fahrzeug mit einer Steuervorrichtung nach Anspruch 2, die so geschaltet ist, daß sie die Arbeitsweise eines die Geschwindigkeit verändernden Gerätes an Bord des Fahrzeuges steuert, wobei die Signalquelle einen Wandler aufweist, um Signale zu erzeugen, die repräsentativ sind für die wenigstens eine physikalische Variable, die der Betriebsökonomie des Fahrzeuges zugeordnet ist und wobei der Computer so programmiert ist, daß er aus den Signalen eine Einstellung des die Geschwindigkeit ändernden Gerätes optimiert auf die Wirtschaftlichkeit berechnet und die Hemmstelle entsprechend im Bewegungsbereich des Steuergliedes anordnet.

4. Fahrzeug nach Anspruch 3, bei welchem das Steuerglied ein durch einen Fuß zu betätigendes Beschleunigungspedal (4) aufweist, und das die Geschwindigkeit ändernde Gerät eine Drossel ist, die die Brennstoffzufuhr nach einer Brennkraftmaschine steuert, wobei die Einstellung des die Geschwindigkeit ändernden Gerätes im Hinblick auf einen optimalen Brennstoffverbrauch berechnet ist (Fig. 1, 2 und 3).

5. Fahrzeug nach Anspruch 3 oder 4 welches eine Schalter (22) aufweist, der mit dem Computer (1) verbunden ist, und bewirkt, daß der Computer das Steuerglied an einer Stelle hält, in der es durch die Bedienungsperson eingestellt wurde.

6. Fahrzeug nach Anspruch 3 oder 4, bei welchem der Computer außerdem den Motor derart erregt, daß eine motorische Bewegung des Steuergliedes erfolgt, wobei eine Parameter anzeigende Signalvorrichtung Mittel besitzt, um die Geschwindigkeit des Fahrzeuges festzustellen und die Steuervorrichtung einen Schalter aufweist, der mit dem Computer verbunden ist und den Computer veranlaßt, die Lage des Steuergliedes so zu ändern, daß die Fahrzeuggeschwindigkeit auf einem konstanten Wert gehalten wird.

7. Flugzeug mit einer Steuervorrichtung nach Anspruch 2, bei welcher die Steuervorrichtung so geschaltet ist, daß sie die Gashebeleinstellung des Triebwerks des Flugzeugs steuert und die Signalquelle ein Signal liefert, welches die Flug-

höhe des Flugzeugs anzeigt, wobei der Computer so programmiert ist, daß er die Lage der Hemmstelle in Abhängigkeit von der Höhe ändert.

8. Flugzeug nach Anspruch 7, bei welchem die den Parameter anzeigende Signalquelle außerdem ein Signal liefert, welches anzeigt, daß das Flugzeug gelandet ist, wobei der Computer so programmiert ist, daß er auf das Flugzeuglandesignal anspricht, indem die Erregung des Motors erhöht wird, um die Möglichkeit zu verringern, daß das Steuerglied aus Versehen aus der Stellung geringer Leistung verschoben wird.

9. Flugzeug nach den Ansprüchen 7 oder 8, bei welchem der Computer einen an einen Autopilotcomputer an Bord des Flugzeuges angeschlossenen Eingang besitzt und alternative gemäß den Signalen betätigbar ist, die vom Autopilotcomputer empfangen werden, um den Motor derart zu erregen, daß eine Bewegung des beweglichen Teiles des Steuergliedes bewirkt wird.

10. Flugzeug nach einem der Ansprüche 7 bis 9, mit einem vom Piloten steuerbaren Schalter, der vom Steuerglied getragen wird, wobei eine Schaltsignalempfangsvorrichtung (108) an der Steuervorrichtung fixiert ist und eine drahtlose Übertragungsvorrichtung (104, 105, 107) Signale übermitteln, die den Zustand des Schalters der Empfangsvorrichtung übermitteln (Fig. 4).

Revendications

1. Procédé pour le fonctionnement d'un dispositif de commande comprenant un élément de commande mobile manuellement (5, 35, 55, 200, 301), un moyen de sortie (2, 14, 33, 52, 302, 402) pour exercer une action de commande sur un équipement associé en dépendance du mouvement de l'élément de commande, et un moteur électrique pas à pas (6, 31, 51, 306) entraîné par ordinateur, couplé à l'élément de commande, le procédé comprenant la fourniture à l'ordinateur (1, 101, 318, 406) d'une information sur la position de l'élément de commande dans sa plage de mouvement et utilisant l'ordinateur pour commander la mise sous tension du moteur de manière à produire une augmentation sensible de la résistance opposée au mouvement manuel de l'élément de commande par l'intermédiaire d'au moins une position d'arrêt dans la plage, le procédé étant caractérisé en ce que l'ordinateur reçoit en outre une information concernant une ou plusieurs variables physiques affectant l'équipement associé et l'ordinateur utilise ladite information pour calculer l'emplacement de la position d'arrêt de façon que cet emplacement change pendant l'emploi du dispositif de commande en fonction des variations de l'une ou de plusieurs des variables.

2. Dispositif de commande pour commander le fonctionnement d'un équipement associé, par exemple un appareil à bord d'un avion ou d'un véhicule routier, le dispositif de commande comprenant un élément de commande mobile manuellement (5, 35, 55, 200, 301), un moteur électrique pas à pas (6, 31, 51, 306) accouplé à

l'élément de commande, un capteur (18, 102, 12, 33, 52, 302) pour détecter la position de l'élément de commande à l'intérieur de sa plage globale de mouvement, un ordinateur (1, 101, 318, 406) connecté au capteur et au moteur pas à pas pour mettre le moteur sous tension et produire une augmentation sensible de la résistance opposée au mouvement manuel de l'élément de commande par l'intermédiaire d'au moins une position d'arrêt discrète de l'élément de commande, et un moyen de sortie (2, 14, 33, 52, 302, 402) pour transmettre une action de commande à l'équipement associé en fonction de la position de l'élément de commande, caractérisé en ce que le dispositif de commande, caractérisé en ce que le dispositif de commande comprend en outre un moyen de fourniture de signaux (103) pour fournir à l'ordinateur (1, 101, 318, 406) des signaux représentatifs d'au moins une variable physique affectant l'équipement associé, et l'ordinateur est programmé pour répondre aux signaux afin de faire varier l'emplacement de la position d'arrêt en fonction de la variable physique.

3. Véhicule motorisé comprenant un dispositif de commande selon la revendication 1, le dispositif de commande étant monté de manière à commander le fonctionnement d'un appareil à vitesse variable à bord du véhicule, le moyen de fourniture de signaux comprenant un moyen de transducteur pour former des signaux représentant au moins une variable physique qui est liée à l'économie de fonctionnement du véhicule et l'ordinateur est programmé de manière à calculer à partir des signaux une valeur de réglage de l'appareil à vitesse variable, optimisée pour ladite économie et pour positionner à l'avenant la position d'arrêt dans la plage de mouvement de l'élément de commande.

4. Véhicule selon la revendication 3, dans lequel l'élément de commande comprend une pédale (4) d'accélérateur actionnée au pied, l'appareil à vitesse variable est un papillon commandant la fourniture de carburant à un moteur à combustion interne, la valeur de réglage de l'appareil à vitesse variable est calculée pour obtenir une économie optimum en carburant (figures 1, 2 et 3).

5. Véhicule selon la revendication 3 ou la revendication 4, comprenant un moyen de commutateur (22) relié à l'ordinateur (1) et pouvant fonctionner pour que l'ordinateur maintienne l'élément de commande à une position dans laquelle il a été réglé par l'opérateur.

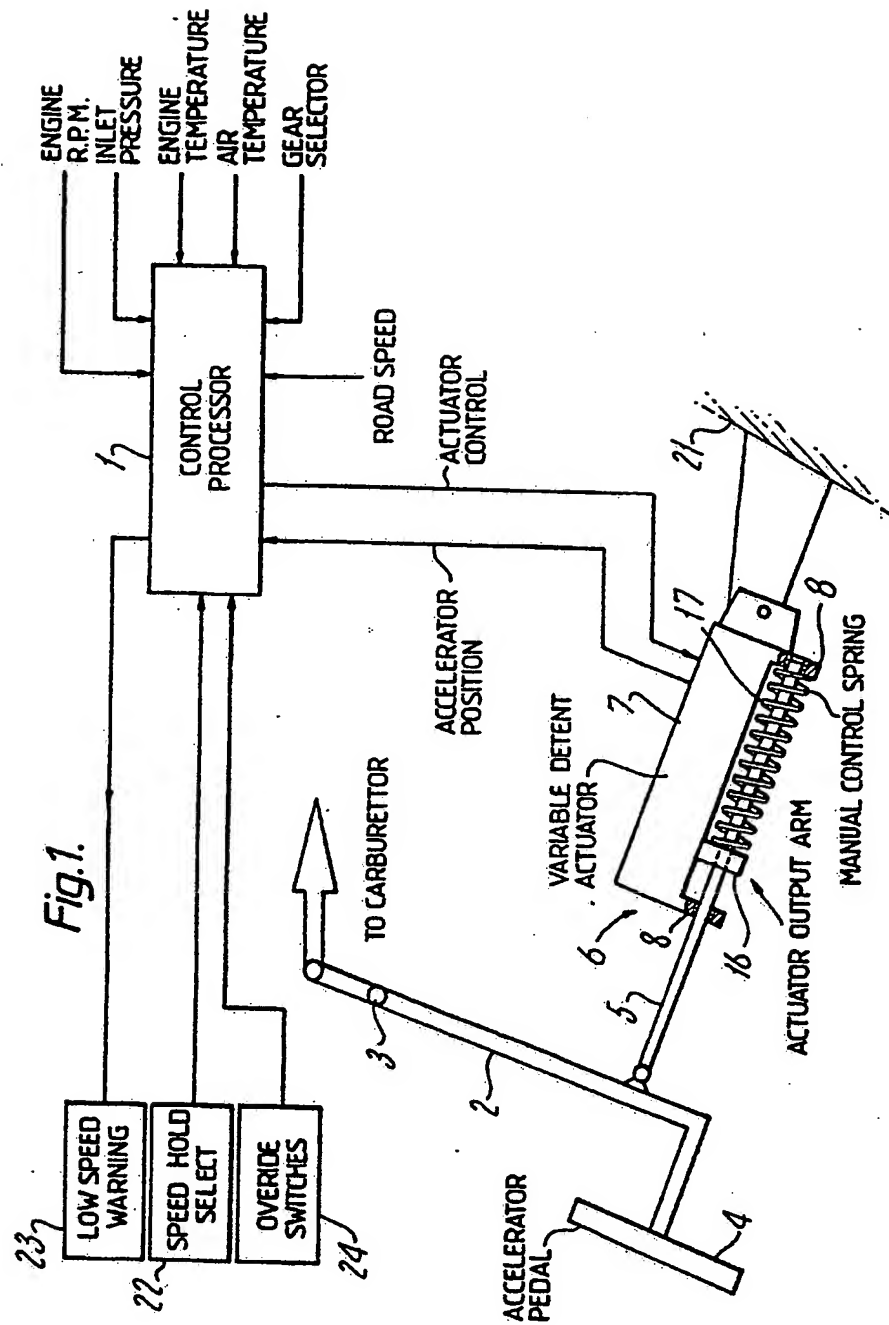
6. Véhicule selon la revendication 3 ou la revendication 4, dans lequel l'ordinateur peut en outre fonctionner pour mettre le moteur sous tension afin de produire un mouvement motorisé de l'élément de commande, le moyen de fourniture de signaux représentatifs d'un paramètre comprend un moyen pour détecter la vitesse du véhicule, et le dispositif de commande comporte un moyen de commutateur relié au moyen d'ordinateur et pouvant fonctionner pour que le moyen d'ordinateur fasse varier la position de l'élément de commande afin de maintenir la vitesse du véhicule à une valeur constante.

7. Avion comprenant un dispositif de commande selon la revendication 2, le dispositif de commande étant monté de manière à commander le réglage du papillon d'un moteur de l'avion, le moyen de fourniture de signaux pouvant fonctionner pour fournir un signal représentatif de l'altitude de l'avion, et l'ordinateur étant programmé pour faire varier l'emplacement de la position d'arrêt en fonction de l'altitude.

8. Avion selon la revendication 7, le moyen de fourniture de signaux représentatifs d'un paramètre pouvant en outre fonctionner pour fournir un signal indiquant que l'avion a atterri, et l'ordinateur étant programmé pour répondre au signal d'atterrissage de l'avion en augmentant l'excitation du moteur de manière à réduire la possibilité que l'élément de commande ne se décarte par inadvertance de la position de réglage à papillon bas.

9. Avion selon la revendication 7 ou la revendication 8, dans lequel l'ordinateur comprend une entrée reliée à un ordinateur de pilotage automatique à bord de l'avion et est alternativement mise en oeuvre en réponse à des signaux provenant de l'ordinateur de pilotage automatique pour mettre le moteur sous tension d'une façon qui provoque le déplacement de sa partie à mécanisme et de l'élément de commande.

10. Avion selon la revendication 7 à 9, comprenant des commutateurs actionnables par le pilote, lesdits commutateurs étant associés à l'élément de commande des moyens de réception (108) des signaux de commande associés au dispositif de contrôle, et des moyens de transmission sans fil (104, 105, 107) pour la transmission des signaux correspondant à l'état desdits commutateurs aux moyens de réception (figure 4).



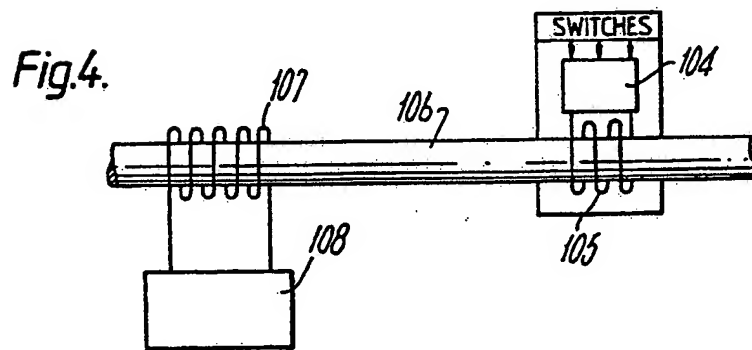
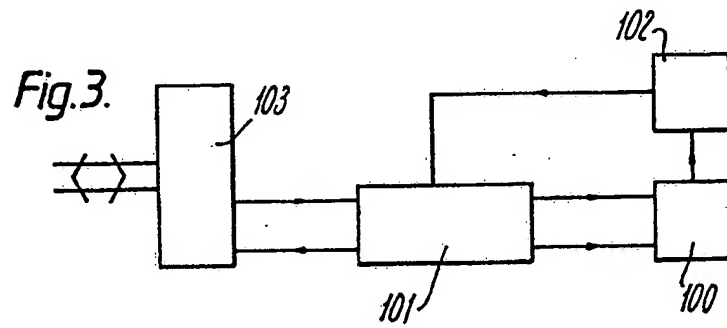
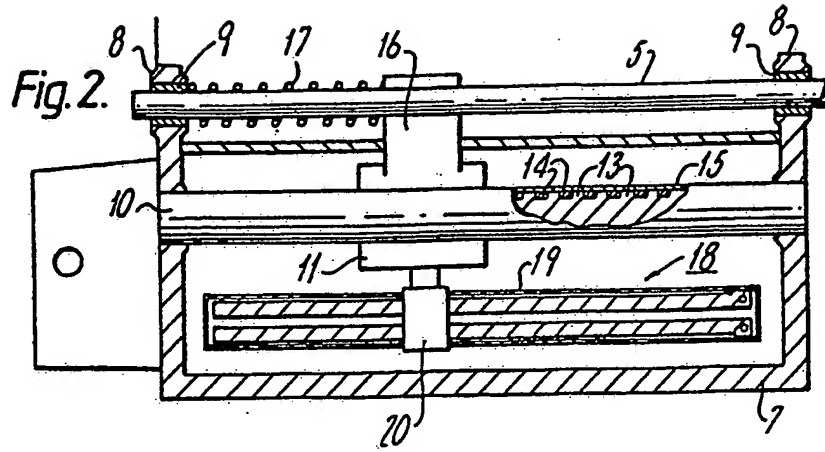


Fig.5.

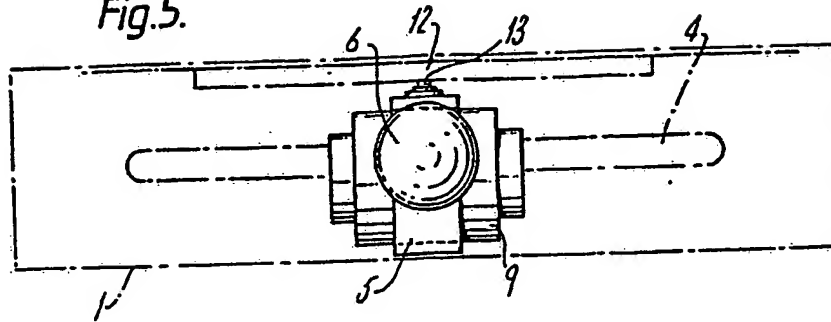


Fig.6.

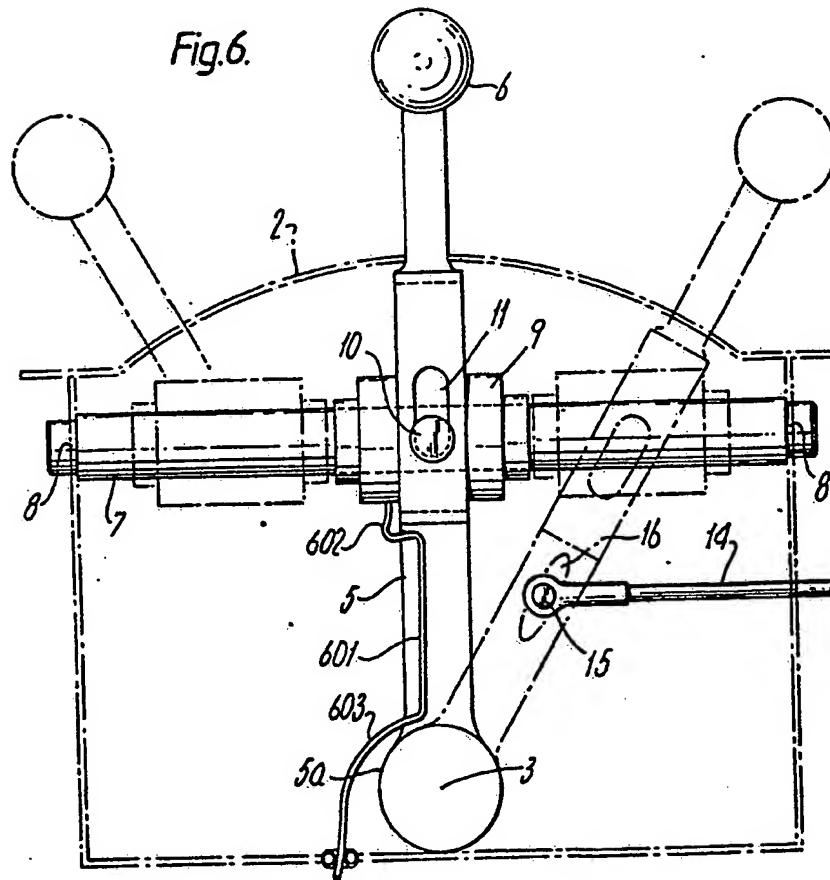


Fig.7

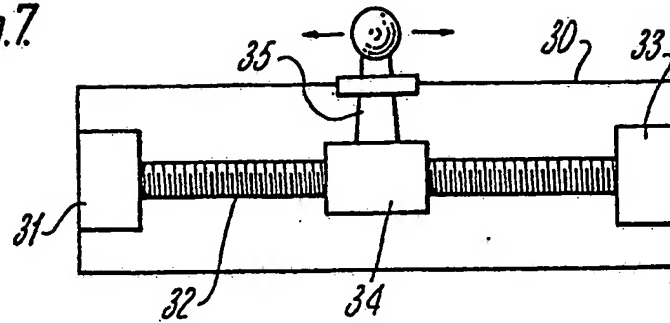


Fig. 8.

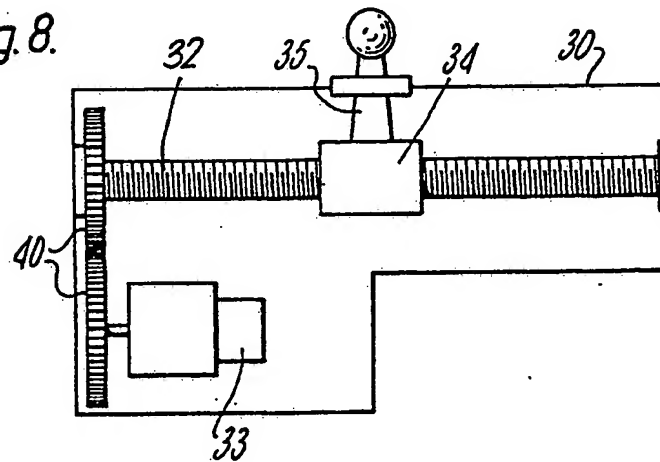


Fig.9

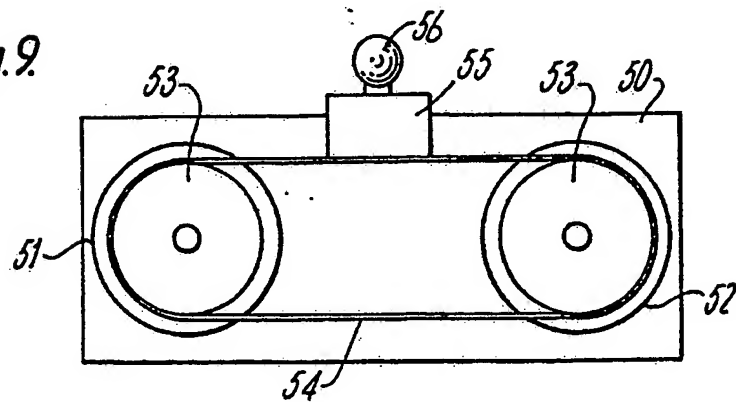


Fig.10.

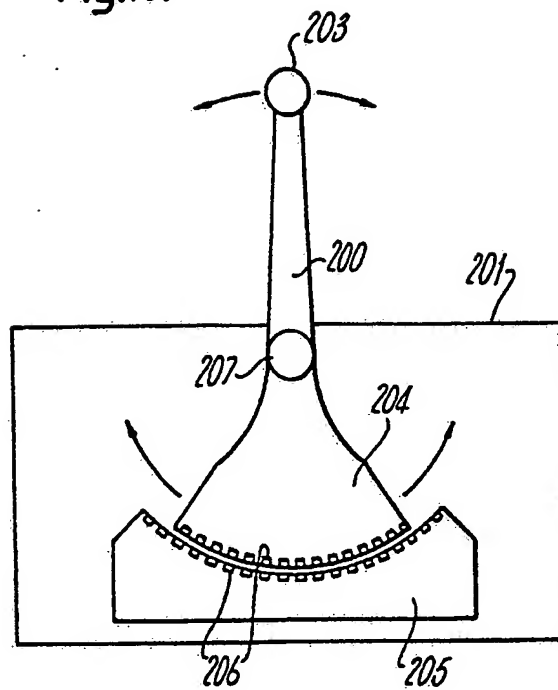


Fig.11.

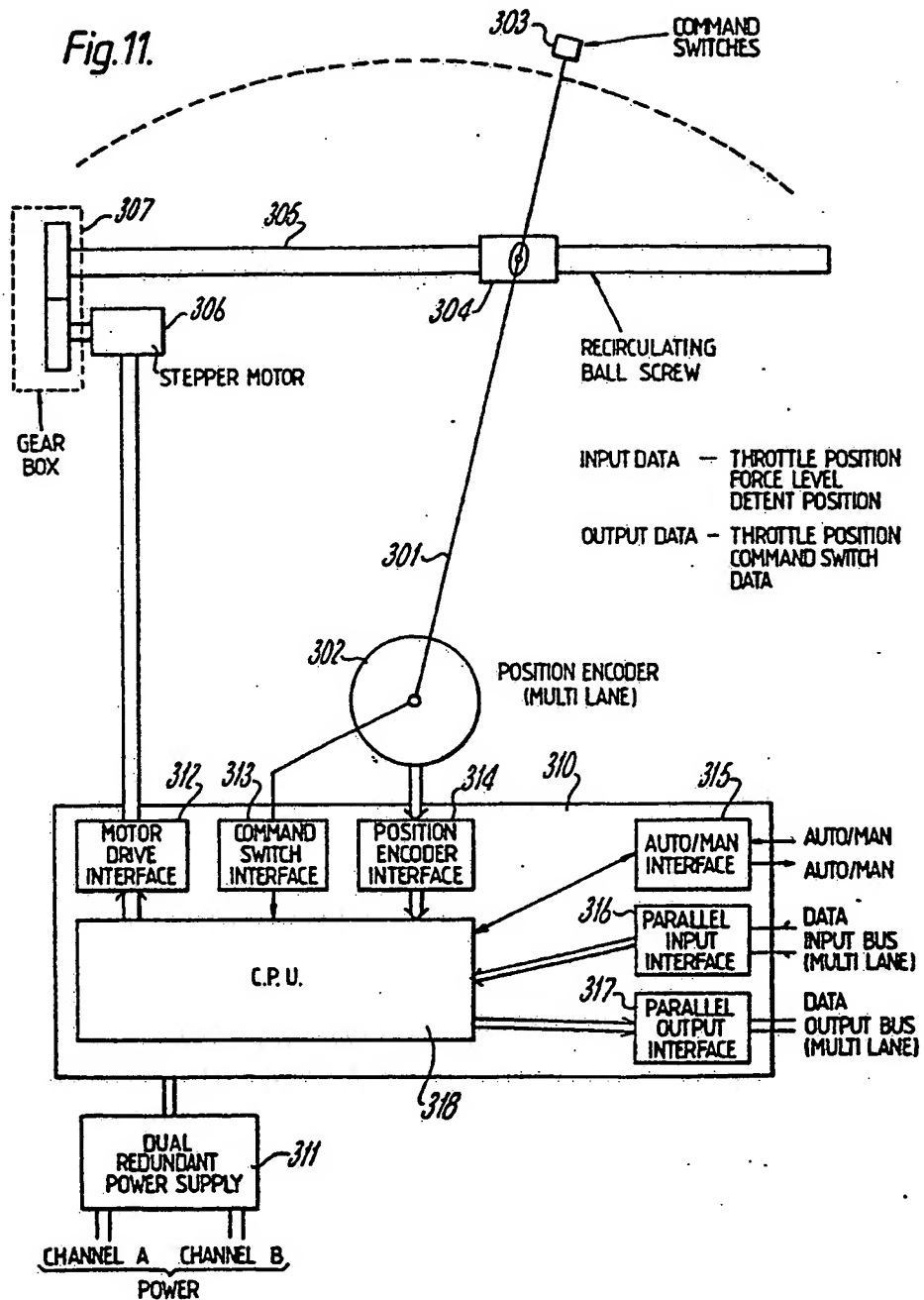


Fig.12.

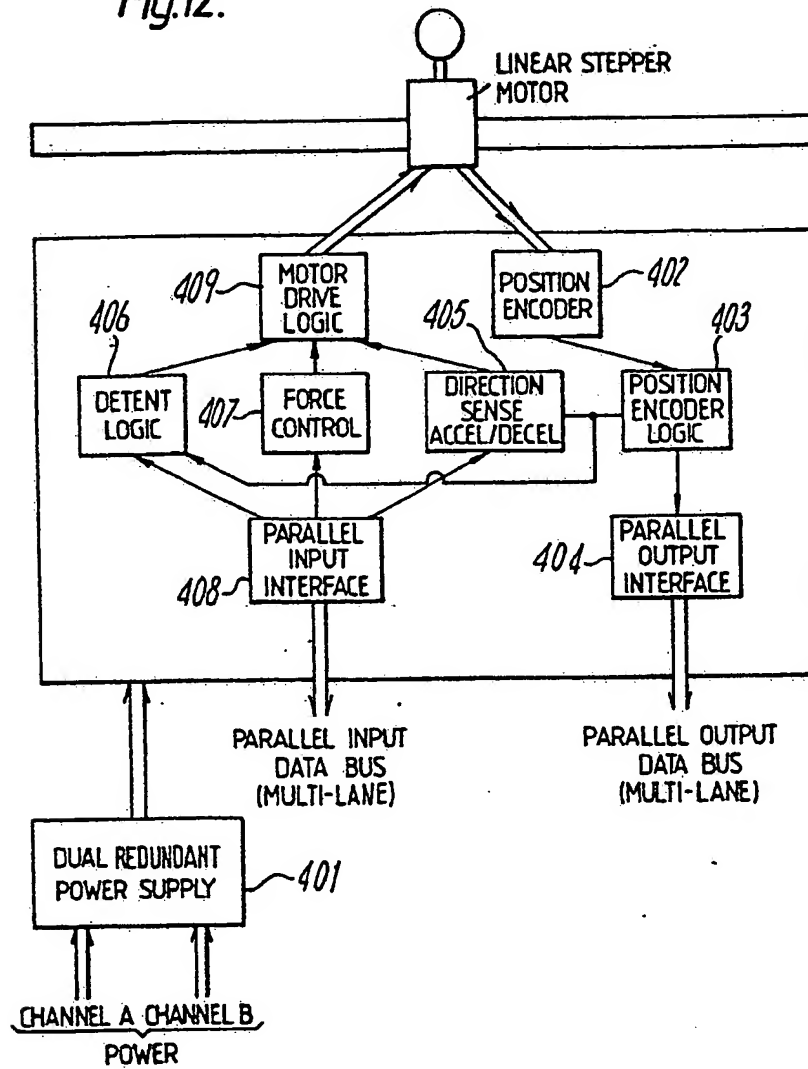


Fig. 13.

